Contact: Václav Sadílek E-mail: kelidas@centrum.cz Office: C502

1 Direct Stiffness Method

1.1 Continuous beam

Use Direct Stiffness Method to calculate the internal forces of the continuous beam with rectangular cross-section 0.4×0.6 m and draw them. The beam is at the both end fixed (clamped, built-in, made-in, restrained) and in the middle of the span there is a mobile support. The loading of the beam consists of a single force (point load) 20 kN at angle 60°, uniform load 5 kNm⁻¹ in length 2 m and a single moment load 10 kNm (2 meters from the support b). See Fig. 1. The material of this beam has modulus of elasticity E = 36 GPa.



Fig. 1: Task scheme

We can write basic equation system:

$$[\mathbf{k}] \cdot \{\mathbf{r}\} = \{\mathbf{F}\} \tag{1}$$

where [k] = stiffness matrix of the bar system, $\{F\}$ = force vector, $\{r\}$ = displacement vector. Cross-sectional area

$$A = b \cdot h = 0.4 \cdot 0.6 = 0.24 \,\mathrm{m^2}$$

The second moment of area

$$I_{\rm y} = \frac{1}{12} \cdot b \cdot h^3 = \frac{1}{12} \cdot 0.4 \cdot 0.6^3 = 0.0072 \,\mathrm{m}^4$$

1. We label nodes a, b, c and calculate degree of kinematic indeterminacy $n_{\rm k}$ (degrees of freedom). The kinematic degree of freedom is the number of independent joint displacements (rotations and translations). In a planar task every joint has three degrees of freedom (u_i, w_i, φ_i) . One translation in the global X-direction, one translation in the Y-direction and one rotation about Z-axis (counter clockwise). Displacement vector $\{r\}$ consists of unknown degrees of freedom, in our case $\{r\} = \{u_{\rm b}, \varphi_{\rm b}\}^{\rm T} \Rightarrow n_{\rm k} = 2$. We know displacement of some nodes (e.g. $u_{\rm a} = 0$ node a is supported in X-direction, etc.). See Fig. 2.

$$\{\boldsymbol{r}\} = \left\{ \begin{array}{c} u_{\rm b} \\ \varphi_{\rm b} \end{array} \right\} \tag{2}$$

$$[0, 0, 0) \qquad (u_{\mathbf{b}}, 0, \varphi_{\mathbf{b}}) \qquad (0, 0, 0)$$

Fig. 2: Labels of nodes and degrees of freedom

2. We divide our continuous beam into two bars (members) a-b and b-c see Fig. 3.

$$F_{\rm x} = F \cdot \cos(\alpha) = 20 \cdot \cos(60^{\circ}) = 10 \,\text{kN}$$
$$F_{\rm z} = F \cdot \sin(\alpha) = 20 \cdot \sin(60^{\circ}) = 17.32 \,\text{kN}$$



Fig. 3: Division of the beam into two bars a-b and b-c

To calculate stiffness matrix of members – we use tab. 11.3a, because bars a-b and b-c are fixed supported on both ends. The matrix is symmetric. The stiffness matrices $[\mathbf{k}_{ab}]$ and $[\mathbf{k}_{bc}]$ are identical by reason of the same bar length, cross-sectional area, the second moment of area and modulus of elasticity.

$$\begin{bmatrix} \mathbf{k}_{ab} \end{bmatrix} = \begin{bmatrix} \mathbf{k}_{bc} \end{bmatrix} = \begin{bmatrix} 1.44 & 0 & 0 & -1.44 & 0 & 0 \\ 0 & 0.0144 & -0.0432 & 0 & -0.0144 & -0.0432 \\ 0 & -0.0432 & 0.1728 & 0 & 0.432 & 0.0864 \\ -1.44 & 0 & 0 & 1.44 & 0 & 0 \\ 0 & -0.0144 & 0.0432 & 0 & 0.0144 & 0.0432 \\ 0 & -0.0432 & 0.0864 & 0 & 0.0432 & 0.1728 \end{bmatrix} \begin{bmatrix} \cdot 10^9 \frac{N}{m} \end{bmatrix}$$

When we solve only displacement vector $\{r\}$, we need only some matrix elements of [k] and $\{\bar{R}\}$. These elements are highlighted below this text and in the next step 3.

$$[k_{
m ab}] =$$

 $[k_{
m bc}] =$

$$\begin{pmatrix} u_{\rm b}(1) & w_{\rm b}(0) & \varphi_{\rm b}(2) & u_{\rm c}(0) & w_{\rm c}(0) & \varphi_{\rm c}(0) \\ 1.44 & 0 & 0 & -1.44 & 0 & 0 \\ 0 & 0.0144 & -0.0432 & 0 & -0.0144 & -0.0432 \\ 0 & -0.0432 & 0.1728 & 0 & 0.0432 & 0.0864 \\ -1.44 & 0 & 0 & 1.44 & 0 & 0 \\ 0 & -0.0144 & 0.0432 & 0 & 0.0144 & 0.0432 \\ 0 & -0.0432 & 0.0864 & 0 & 0.0432 & 0.1728 \end{pmatrix} \begin{vmatrix} u_{\rm b}(1) & w_{\rm b}(0) \\ \varphi_{\rm b}(2) & u_{\rm c}(0) \\ u_{\rm c}(0) & w_{\rm c}(0) \\ \varphi_{\rm c}(0) \end{vmatrix}$$

Using highlighted matrix elements, we will assemble global stiffness matrix of the system

ī

$$[\mathbf{k}] = \begin{pmatrix} 1.44 \cdot 10^9 & 0 \\ +1.44 \cdot 10^9 & +0 \\ \hline 0 & 1.728 \cdot 10^8 \\ +0 & +1.728 \cdot 10^8 \end{pmatrix} = \begin{pmatrix} 2.88 \cdot 10^9 & 0 \\ 0 & 3.456 \cdot 10^8 \end{pmatrix}$$
(3)

3. Force vector $\{F\}$ is equal to nodal force vector $\{S\}$ take away primary force vector $\{\bar{R}\}$ (the strip above R signify that the vector is primary vector). In our case there is no nodal load this implies $\{S\} = \emptyset$.

$$\{F\} = \{S\} - \{\bar{R}\}$$

$$\tag{4}$$

Primary vector $\left\{ \bar{R}_{ab} \right\}$ of single force - see tab. 11.2c (14.10 row 2)

$$\left\{ \bar{\boldsymbol{R}}_{ab} \right\} = \left\{ \begin{array}{c} \bar{X}_{ab} \\ \bar{Z}_{ab} \\ \bar{M}_{ab} \\ \bar{X}_{ba} \\ \bar{Z}_{ba} \\ \bar{Z}_{ba} \\ \bar{M}_{ba} \end{array} \right\} = \left\{ \begin{array}{c} -5000 \\ -8660 \\ 12990 \\ -5000 \\ -8660 \\ -12990 \end{array} \right\} [N]$$
(5)

Primary vector $\{\bar{R}_{bc}^1\}$ of uniform load in length 2 m - see tab. 14.10(row 11 – formulas are given in the second column)

$$\left\{ \bar{\boldsymbol{R}}_{bc}^{1} \right\} = \left\{ \begin{array}{c|c} \bar{X}_{bc}^{1} \\ \bar{Z}_{bc}^{1} \\ \bar{M}_{bc}^{1} \\ \bar{X}_{cb}^{1} \\ \bar{X}_{cb}^{1} \\ \bar{Z}_{cb}^{1} \\ \bar{M}_{cb}^{1} \end{array} \right\} = \left\{ \begin{array}{c|c} 0 & 0 \\ -9074 & -\frac{q \cdot a}{2 \cdot l^{3}} \left[2 \cdot l \cdot \left(l^{2} - a^{2} \right) + a^{3} \right] \\ 6111 & \frac{q \cdot a^{2}}{12 \cdot l^{2}} \left(6 \cdot b^{2} + 3 \cdot a \cdot b + a \cdot l \right) \\ 0 & 0 \\ -926 & -\frac{q \cdot a^{3}}{2 \cdot l^{3}} \left(l + b \right) \\ -1667 & -\frac{q \cdot a^{3}}{12 \cdot l^{3}} \left(3 \cdot b - l \right) \end{array} \right\} \left[N$$

Primary vector $\{\bar{R}_{bc}^2\}$ of single moment load - see tab. 11.2d (14.10 row 9)

$$\left\{ \bar{\boldsymbol{R}}_{bc}^{2} \right\} = \left\{ \begin{array}{c} \bar{X}_{bc}^{2} \\ \bar{Z}_{bc}^{2} \\ \bar{M}_{bc}^{2} \\ \bar{X}_{cb}^{2} \\ \bar{Z}_{cb}^{2} \\ \bar{M}_{cb}^{2} \end{array} \right\} = \left\{ \begin{array}{c} 0 \\ -2222 \\ 0 \\ 0 \\ 2222 \\ 3333 \end{array} \right\} [N]$$

Primary vector $\left\{ ar{m{R}}_{
m bc}
ight\}$

$$\{\bar{\boldsymbol{R}}_{bc}\} = \{\bar{\boldsymbol{R}}_{bc}^{1}\} + \{\bar{\boldsymbol{R}}_{bc}^{2}\} = \begin{cases} \bar{\boldsymbol{X}}_{bc} \\ \bar{\boldsymbol{Z}}_{bc} \\ \bar{\boldsymbol{M}}_{bc} \\ \bar{\boldsymbol{X}}_{cb} \\ \bar{\boldsymbol{Z}}_{cb} \\ \bar{\boldsymbol{M}}_{cb} \end{cases} = \begin{cases} 0 \\ -11296 \\ 6111 \\ 0 \\ 1296 \\ 1666 \end{cases} [N]$$
(6)

Force vector $\{F\}$ see eq. 4. We use only elements in relation to unknown displacements $u_{\rm b}$ and $\varphi_{\rm b}$.

$$\{\mathbf{F}\} = \{\emptyset\} - \left\{\begin{array}{c} -5000 + 0\\ -12990 + 6111 \end{array}\right\} = \left\{\begin{array}{c} 5000\\ 6879 \end{array}\right\}$$

4. Solution of vector of unknown displacements $\{r\}$

$$\begin{pmatrix} 2.88 \cdot 10^9 & 0 \\ 0 & 3.456 \cdot 10^8 \end{pmatrix} \cdot \begin{cases} u_{\rm b} \\ \varphi_{\rm b} \end{cases} = \begin{cases} 5000 \\ 6879 \end{cases}$$
$$\begin{cases} u_{\rm b} \\ \varphi_{\rm b} \end{cases} = \begin{cases} 1.736 \cdot 10^{-6} \\ 19.905 \cdot 10^{-6} \end{cases} \begin{bmatrix} m \\ rad \end{bmatrix}$$

5. Solution of end reaction vector. $\{\widehat{R}_{bc}\}$ is the secondary force vector. To solve this equation system we need only highlighted columns, which are in relation to displacements differing from 0.

$$\{\boldsymbol{R}_{\rm ab}\} = \{\bar{\boldsymbol{R}}_{\rm ab}\} + \{\bar{\boldsymbol{R}}_{\rm ab}\} = \{\bar{\boldsymbol{R}}_{\rm ab}\} + [\boldsymbol{k}_{\rm ab}] \cdot \{\boldsymbol{r}_{\rm ab}\}$$
(7)

$$\{\boldsymbol{R}_{\rm bc}\} = \{\bar{\boldsymbol{R}}_{\rm bc}\} + \{\hat{\boldsymbol{R}}_{\rm bc}\} = \{\bar{\boldsymbol{R}}_{\rm bc}\} + [\boldsymbol{k}_{\rm bc}] \cdot \{\boldsymbol{r}_{\rm bc}\}$$
(8)

$$\{\boldsymbol{R}_{ab}\} = \begin{cases} X_{ab} \\ Z_{ab} \\ M_{ab} \\ X_{ba} \\ Z_{ba} \\ M_{ba} \end{cases} = \begin{cases} -5000 \\ -8660 \\ 12990 \\ -5000 \\ -8660 \\ -12990 \end{cases} + \left(\begin{array}{ccccc} 1.44 & 0 & 0 & -1.44 & 0 & 0 \\ 0 & 0.0144 & -0.0432 & 0 & -0.0144 & -0.0432 \\ 0 & -0.0432 & 0.1728 & 0 & 0.0432 & 0.0864 \\ -1.44 & 0 & 0 & 1.44 & 0 & 0 \\ 0 & -0.0144 & 0.0432 & 0 & 0.0144 & 0.0432 \\ 0 & -0.0432 & 0.0864 & 0 & 0.0432 & 0.1728 \end{array} \right) \begin{cases} 0 \\ 0 \\ 1.736 \\ 0 \\ 1.736 \\ 0 \\ 19.905 \end{array} \right) \cdot 10^3$$

There is value 10^3 that we obtain by multiplying stiffness matrix multiplier 10^9 and displacement vector multiplier 10^{-6} .

$$\{\boldsymbol{R}_{ab}\} = \begin{cases} X_{ab} \\ Z_{ab} \\ M_{ab} \\ X_{ba} \\ Z_{ba} \\ M_{ba} \end{cases} = \begin{cases} -7500.16 \\ -9519.90 \\ 14709.80 \\ -2500.16 \\ -7800.00 \\ -9550.00 \end{cases} [N, Nm]$$
$$\{\boldsymbol{R}_{bc}\} = \begin{cases} X_{bc} \\ Z_{bc} \\ M_{bc} \\ X_{cb} \\ Z_{cb} \\ M_{cb} \end{cases} = \begin{cases} 2500 \\ -12156 \\ 9550 \\ -2500 \\ 2156 \\ 3385 \end{cases} [N, Nm]$$



Fig. 4: Force equilibrium on the bar $[\rm kN,\,\rm kNm]$

6. Check force equilibrium on the bar – see Fig. 4.

Bar a - b

$$\sum F_x = 0: -7.50 + 10 - 2.50 = 0$$

$$\sum F_z = 0: -9.52 + 17.32 - 7.8 = 0$$

$$\sum M_a = 0: 14.71 - 17.32 \cdot 3 - (-7.8) \cdot 6 + (-9.55) = 0$$
Bar b. a

Bar b-c

$$\sum F_x = 0: 2.50 + (-2.50) = 0$$
$$\sum F_z = 0: -12.16 + 5 \cdot 2 + 2.16 = 0$$
$$\sum M_b = 0: 9.55 - \frac{5 \cdot 2^2}{2} + 10 + 3.39 - 2.16 \cdot 6 = 0$$

7. Check force equilibrium in nodes a, b and c – see Fig. 5.

Fig. 5: Force equilibrium in nodes a, b and c $[\rm kN,\,\rm kNm]$

Node a

$$\sum F_{xa} = 0 : R_{ax} - (-7.50) = 0 \Rightarrow R_{ax} = -7.50 \text{ kN}$$
$$\sum F_{za} = 0 : R_{az} - (-9.52) = 0 \Rightarrow R_{az} = -9.52 \text{ kN}$$
$$\sum M_a = 0 : M_c - 14.71 = 0 \Rightarrow M_a = 14.71 \text{ kNm}$$

Node b

$$\sum F_{xb} = 0 : -2.50 + 2.50 = 0$$

$$\sum F_{zb} = 0 : R_{bz} - 7.80 - 12.16 = 0 \Rightarrow R_{bz} = 19.96 \text{ kN}$$

$$\sum M_b = 0 : -9.55 + 9.55 = 0$$

Node c

$$\sum F_{xc} = 0 : R_{cx} - (-2.50) = 0 \Rightarrow R_{cx} = -2.50 \text{ kN}$$
$$\sum F_{zc} = 0 : R_{cz} + 2.16 = 0 \Rightarrow R_{cz} = -2.16 \text{ kN}$$
$$\sum M_c = 0 : M_c - 3.39 = 0 \Rightarrow M_c = 3.39 \text{ kNm}$$

8. Internal forces diagrams (N – normal force, V – shear force, M – bending moment) – see Fig. 6.



1.2 2D Frame

Use Direct Stiffness Method to calculate the internal forces of the frame and draw them. See Fig. 7. Material parameters: modulus of elasticity E = 24 GPa and coefficient of thermal expansion $\alpha_t = 10^{-5}$ K⁻¹.



Fig. 7: Left:2D frame, right: model scheme.

Cross-sectional area

$$A_{\rm ab} = b \cdot h = 0.4 \cdot 0.6 = 0.24 \,\mathrm{m}^2$$
$$A_{\rm bc} = A_{\rm cd} = b \cdot h = 0.4 \cdot 0.3 = 0.12 \,\mathrm{m}^2$$

The second moment of area

$$I_{y}^{ab} = \frac{1}{12} \cdot b \cdot h^{3} = \frac{1}{12} \cdot 0.4 \cdot 0.6^{3} = 0.0072 \text{ m}^{4}$$
$$I_{y}^{bc} = I_{y}^{cd} = \frac{1}{12} \cdot b \cdot h^{3} = \frac{1}{12} \cdot 0.4 \cdot 0.3^{3} = 0.0009 \text{ m}^{4}$$

1. degree of kinematic indeterminacy (number of degrees of freedom)

$$n_k = 5$$

2. displacement vector and nodal force vector

$$\{r\} = \begin{cases} u_{\rm b} \\ w_{\rm b} \\ \varphi_{\rm b} \\ u_{\rm c} \\ w_{\rm c} \end{cases} \qquad \{S\} = \begin{cases} 0 \\ 0 \\ -4\,000 \\ -10\,000 \\ 6\,000 \end{cases} [\rm N, Nm] \qquad (9)$$



Fig. 8: a) global coordinate system xz, b) local coordinate system of the general bar x^*z^* and determination of angle γ (clockwise) c) angle γ of our bars a-b, b-c and c-d

3. stiffness matrix in global coordinate system

(a) bar a-b — tab. 11.4a, angle $\gamma=270^\circ \Rightarrow s=-1, c=0$

$[k_{ m ab}] =$										
	$u_{\rm a}$	$w_{\rm a}$	φ_{a}	$u_{ m b}$	$w_{ m b}$	$arphi_{ m b}$				
(-32.4	0	-64.8		$u_{\rm a}$		
				0	-1440	0		$w_{\rm a}$		
				64.8	0	86.4		$arphi_{ m a}$	[106N]	
				32.4	0	64.8		$u_{\rm b}$	$\begin{bmatrix} 10 \\ \overline{m} \end{bmatrix}$	
				0	1440	0		$w_{\rm b}$		
				64.8	0	172.8)	φ_{b}		

(b) bar b-c – local coordinate system of the bar is same as global coordinate of the system — we can use tab. 11.3b

$[m{k}_{ m bc}] =$	$[m{k}^{\star}_{ m bc}] =$							
	$u_{ m b}$	$w_{ m b}$	$arphi_{ m b}$	$u_{\rm c}$	$w_{\rm c}$	$\varphi_{ m c}$		
(1440	0	0	-1440	0	0	$u_{\rm b}$	
	0	8.1	-16.2	0	-8.1	0	w_{b}	
	0	-16.2	32.4	0	16.2	0	$arphi_{ m b}$	$[.10^{6} \frac{N}{N}]$
	-1440	0	0	1440	0	0	$u_{\rm c}$	$\begin{bmatrix} 10 \\ \overline{m} \end{bmatrix}$
	0	-8.1	16.2	0	8.1	0	$w_{ m c}$	
	0	0	0	0	0	0	$\int \varphi_{\rm c}$	

(c) bar c-d — tab. 11.4d, angle $\gamma = 53.130^\circ \Rightarrow s = \frac{4}{5}, c = \frac{3}{5}$

$$\begin{bmatrix} \mathbf{k}_{cd} \end{bmatrix} = \\ \begin{bmatrix} u_c & w_c & \varphi_c & u_d & w_d & \varphi_d \\ 207.36 & 276.48 & . & . & . \\ 276.48 & 368.64 & . & . & . & . \\ 0 & 0 & . & . & . & . \\ -207.36 & -276.48 & . & . & . & . \\ -276.48 & -368.64 & . & . & . & . \\ 0 & 0 & . & . & . & . \\ \end{bmatrix} \begin{bmatrix} u_c & w_c & w_d & \varphi_d & \vdots \\ w_c & \varphi_c & \vdots \\ u_d & \vdots \\ w_d & \vdots \\ w_d & \varphi_d & \vdots \\ \vdots \end{bmatrix}$$

(d) stiffness matrix of the system

$$[k] =$$

$u_{ m b}$	$w_{ m b}$	$arphi_{ m b}$	$u_{ m c}$	$w_{\mathbf{c}}$		
32.4 + 1440	0 + 0	64.8 + 0	-1440	0	$u_{\rm b}$	
0 + 0	1440 + 8.1	0 + (-16.2)	0	-8.1	$w_{ m b}$	
64.8 + 0	0 + (-16.2)	172.8 + 32.4	0	16.2	$\varphi_{\rm b} \left[\cdot 10^6 \frac{\rm N}{\rm m} \right]$	
-1440	0	0	1440 + 207.36	0 + 276.48	$u_{ m c}$	
0	-8.1	16.2	0 + 276.48	8.1 + 368.64	$\int w_{\rm c}$	

$$[k] =$$

$$\begin{pmatrix} u_{\rm b} & w_{\rm b} & \varphi_{\rm b} & u_{\rm c} & w_{\rm c} \\ 1472.4 & 0 & 64.8 & -1440 & 0 \\ 0 & 1448.1 & -16.2 & 0 & -8.1 \\ 64.8 & -16.2 & 205.2 & 0 & 16.2 \\ -1440 & 0 & 0 & 1647.36 & 276.48 \\ 0 & -8.1 & 16.2 & 276.48 & 376.74 \end{pmatrix} \begin{pmatrix} u_{\rm b} \\ w_{\rm b} \\ \varphi_{\rm b} & \left[\cdot 10^6 \frac{\rm N}{\rm m}\right] \\ u_{\rm c} \\ w_{\rm c} \end{pmatrix}$$

- 4. primary vectors of end reactions in local coordinates of the bar
 - (a) bar a-b tab 11.2a (tab. 14.10 row 13)

$$\left\{ \bar{\boldsymbol{R}}_{ab}^{\star} \right\} = \left\{ \begin{array}{c} \bar{X}_{ab}^{\star} \\ \bar{Z}_{ab}^{\star} \\ \bar{M}_{ab}^{\star} \\ \bar{X}_{ba}^{\star} \\ \bar{Z}_{ba}^{\star} \\ \bar{M}_{ba}^{\star} \end{array} \right\} = \left\{ \begin{array}{c} 0 \\ -16 \\ 10.\bar{6} \\ 0 \\ -16 \\ -10.\bar{6} \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(10)

– transformation to the global coordinate system

$$\{\bar{\boldsymbol{R}}_{ab}\} = \begin{cases} \bar{X}_{ab} \\ \bar{Z}_{ab} \\ \bar{M}_{ab} \\ \bar{X}_{ba} \\ \bar{Z}_{ba} \\ \bar{M}_{ba} \end{cases} = \begin{cases} -16 \\ 0 \\ 10.\bar{6} \\ -16 \\ 0 \\ -10.\bar{6} \end{cases} \left[\cdot 10^3 \,\mathrm{N, Nm} \right]$$
(11)

(b) bar b-c — force F_1 : tab. 11.2c (tab. 14.11 row 2) and thermal load t: tab. 11.5b

$$\left\{ \bar{\boldsymbol{R}}_{bc}^{\star} \right\}_{F_{1}} = \left\{ \bar{\boldsymbol{R}}_{bc} \right\}_{F_{1}} = \left\{ \begin{array}{c} 0 \\ -2.75 \\ 1.5 \\ 0 \\ -1.25 \\ 0 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(12)

$$\left\{ \bar{\boldsymbol{R}}_{bc}^{\star} \right\}_{t} = \left\{ \bar{\boldsymbol{R}}_{bc} \right\}_{t} = \left\{ \begin{array}{c} 57.6 \\ -4.32 \\ 8.64 \\ -57.6 \\ 4.32 \\ 0 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(13)

$$\left\{ \bar{\boldsymbol{R}}_{bc}^{\star} \right\} = \left\{ \bar{\boldsymbol{R}}_{bc} \right\} = \left\{ \begin{array}{c} \bar{X}_{bc}^{\star} \\ \bar{Z}_{bc}^{\star} \\ \bar{M}_{bc}^{\star} \\ \bar{X}_{cb}^{\star} \\ \bar{Z}_{cb}^{\star} \\ \bar{M}_{cb}^{\star} \end{array} \right\} = \left\{ \begin{array}{c} 57.6 \\ -7.07 \\ 10.14 \\ -57.6 \\ 3.07 \\ 0 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(14)

(c) bar c-d — similar to simply supported beam

$$\left\{ \bar{\boldsymbol{R}}_{cd}^{\star} \right\} = \left\{ \begin{array}{c} \bar{X}_{cd}^{\star} \\ \bar{Z}_{cd}^{\star} \\ \bar{M}_{cd}^{\star} \\ \bar{X}_{dc}^{\star} \\ \bar{Z}_{dc}^{\star} \\ \bar{Z}_{dc}^{\star} \\ \bar{M}_{dc}^{\star} \end{array} \right\} = \left\{ \begin{array}{c} -4 \\ -3 \\ 0 \\ -4 \\ -3 \\ 0 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(15)

– transformation to the global coordinate system

$$\left\{ \bar{\boldsymbol{R}}_{cd} \right\} = \left\{ \begin{array}{c} \bar{X}_{cd} \\ \bar{Z}_{cd} \\ \bar{M}_{cd} \\ \bar{X}_{dc} \\ \bar{Z}_{dc} \\ \bar{M}_{dc} \end{array} \right\} = \left\{ \begin{array}{c} 0 \\ -5 \\ 0 \\ 0 \\ -5 \\ 0 \end{array} \right\} \left[\cdot 10^3 \,\mathrm{N, Nm} \right]$$
(16)

(d) primary vector of the system

$$\left\{ \bar{\boldsymbol{R}} \right\} = \left\{ \begin{array}{c} -16 + 57.6 \\ 0 + (-7.07) \\ -10.\bar{6} + 10.14 \\ -57.6 + 0 \\ 3.07 + (-5) \end{array} \right\} \cdot 10^{3} = \left\{ \begin{array}{c} 41.6 \\ -7.07 \\ -0.52\bar{6} \\ -57.6 \\ -1.93 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N, Nm} \right]$$
(17)

5. force vector

$$\{\boldsymbol{F}\} = \{\boldsymbol{S}\} - \{\bar{\boldsymbol{R}}\} = \left\{ \begin{array}{c} 0\\0\\-4\\-10\\6 \end{array} \right\} \cdot 10^{3} - \left\{ \begin{array}{c} 41.6\\-7.07\\-0.52\bar{6}\\-57.6\\-1.93 \end{array} \right\} \cdot 10^{3} = \left\{ \begin{array}{c} -41.6\\7.07\\-3.47\bar{3}\\47.6\\7.93 \end{array} \right\} \left[\cdot 10^{3} \,\mathrm{N,Nm} \right] \quad (18)$$

6. equation system

$$[\boldsymbol{k}] \cdot \{\boldsymbol{r}\} = \{\boldsymbol{F}\} \tag{19}$$

$$\begin{pmatrix} 1472.4 & 0 & 64.8 & -1440 & 0 \\ 0 & 1448.1 & -16.2 & 0 & -8.1 \\ 64.8 & -16.2 & 205.2 & 0 & 16.2 \\ -1440 & 0 & 0 & 1647.36 & 276.48 \\ 0 & -8.1 & 16.2 & 276.48 & 376.74 \end{pmatrix} \cdot 10^{6} \cdot \begin{cases} u_{\rm b} \\ w_{\rm b} \\ u_{\rm c} \\ w_{\rm c} \end{cases} = \begin{cases} -41.6 \\ 7.07 \\ -3.47\overline{3} \\ 47.6 \\ 7.93 \end{cases} \cdot 10^{3} \\ 47.6 \\ 7.93 \end{cases} \cdot 10^{3} \\ \begin{cases} u_{\rm b} \\ w_{\rm b} \\ \varphi_{\rm b} \\ u_{\rm c} \\ w_{\rm c} \end{cases} = \begin{cases} 39.171 \\ 4.429 \\ -26.783 \\ 67.736 \\ -27.414 \end{cases} \left[\cdot10^{-6} \,\mathrm{m, rad} \right]$$

7. End reaction vector of the bar

$$\{\boldsymbol{R}_{ab}\} = \{\bar{\boldsymbol{R}}_{ab}\} + \{\hat{\boldsymbol{R}}_{ab}\} = \{\bar{\boldsymbol{R}}_{ab}\} + [\boldsymbol{k}_{ab}]\{\boldsymbol{r}_{ab}\}$$
(20)

$$\left\{ \hat{\mathbf{R}}_{ab} \right\} = \left(\begin{array}{ccccc} \cdot \cdot \cdot & -32.4 & 0 & -64.8 \\ \cdot \cdot \cdot & 0 & -1440 & 0 \\ \cdot \cdot & \cdot & 64.8 & 0 & 86.4 \\ \cdot & \cdot & 0 & 1440 & 0 \\ \cdot & \cdot & \cdot & 64.8 & 0 & 172.8 \end{array} \right) \cdot \left\{ \begin{array}{c} 0 \\ 0 \\ 0 \\ 3.171 \\ 4.429 \\ -26.783 \end{array} \right\} = \left\{ \begin{array}{c} -466.398 \\ -6377.76 \\ -224.20 \\ 0 \\ -24.203 \end{array} \right\} \\ \left\{ \hat{\mathbf{R}}_{bc} \right\} = \left\{ \begin{array}{c} 1440 & 0 & 0 & -1440 & 0 & 0 \\ 0 & 8.1 & -16.2 & 0 & -8.1 & 0 \\ 0 & -16.2 & 32.4 & 0 & 16.2 & 0 \\ 0 & -8.1 & 16.2 & 0 & 8.1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{array} \right) \cdot \left\{ \begin{array}{c} 39.171 \\ 4.429 \\ -26.783 \end{array} \right) \cdot \left\{ \begin{array}{c} \frac{4429}{-26.783} \\ -77.66 \\ -27.414 \\ 0 \\ 0 \end{array} \right\} = \left\{ \begin{array}{c} -41133.6 \\ 691.813 \\ -77.64 \\ -276.48 & 368.64 & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -368.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot \\ -276.48 & -366.64 & \cdot & \cdot & \cdot \\ -276.48 & -366.77.76 & -15533.66 \\ -1646.314 & -8621.752 & 0 \\ 0 & 0 & \cdot & \cdot & \cdot & \cdot \\ 0 & 0 & 0 & \cdot & \cdot & \cdot \\ 0 & 0 & 0 & 0 \\ -1646.398 & -12765.4886 \\ \end{array} \right\}$$



8. Check force equilibrium on the bar – see Fig. 4.

Bar a - b

$$\sum F_x = 0: 6.378 + (-6.378) = 0$$

$$\sum F_z = 0: -15.534 + 8 \cdot 4 - 16.466 = 0$$

$$\sum M_a = 0: 10.891 - 8 \cdot \frac{4^2}{2} + (-12.756) - (-16.466) \cdot 4 = 0$$
Bar b - c

$$\sum F_x = 0: 16.466 + (-16.466) = 0$$

$$\sum F_z = 0: -6.378 + 4 + 2.378 = 0$$

$$\sum M_b = 0: 8.756 - 4 \cdot 1 - 2.378 \cdot 2 = 0$$
Bar c - d

$$\sum F_x = 0: 6.777 + 8 + (-14.777) = 0$$

$$\sum F_z = 0: -3 + 6 + (-3) = 0$$

$$\sum M_b = 0: 6 \cdot 2.5 + (-3) \cdot 5 = 0$$

9. Check force equilibrium in nodes a, b and c and support reactions – see Fig. 5.



Fig. 10: Force equilibrium in nodes a, b, c and d [kN, kNm]

Node a

$$\sum F_{xa} = 0 : R_{ax} - 6.378 = 0 \Rightarrow R_{ax} = 6.378 \text{ kN}$$
$$\sum F_{za} = 0 : R_{az} + (-15.534) = 0 \Rightarrow R_{az} = 15.534 \text{ kN}$$
$$\sum M_{a} = 0 : M_{a} - 10.891 = 0 \Rightarrow M_{a} = 10.891 \text{ kNm}$$

Node b

$$\sum F_{\rm xb} = 0 : -16.466 + 16.466 = 0$$
$$\sum F_{\rm zb} = 0 : 6.378 + (-6.378) = 0$$
$$\sum M_{\rm b} = 0 : 8.756 + 4 + (-12.765) = 0$$

Node c

$$\sum F_{xc} = 0 : -16.466 + 10 + 6.466 = 0$$
$$\sum F_{zc} = 0 : -2.378 + 6 + (-3.622) = 0$$

Node d

$$\sum F_{\rm xd} = 0 : R_{\rm dx} - (-6.466) = 0 \Rightarrow R_{\rm dx} = -6.466 \,\text{kN}$$
$$\sum F_{\rm zd} = 0 : R_{\rm dz} + (-13.622) = 0 \Rightarrow R_{\rm dz} = -13.622 \,\text{kN}$$

10. Internal forces diagrams (N – normal force, V – shear force, M – bending moment) – see Fig. 6.



Fig. 11: Internal forces of the frame – N, V, M $\,$